## Math 542 Spring 2015 — Assignment #5. Due 4/10/2015.

## The *n*-Body Problem

The *n*-body problem is the problem of finding, given the initial positions, masses, and velocities of n bodies, their subsequent motions as determined by classical mechanics, i.e., Newton's laws of motion and Newton's law of gravity.

For given initial data  $\vec{x}_k, k = 1, \ldots, n$  we get the equations

$$m_j \frac{d^2}{dt^2} \vec{x}_j = \gamma \sum_{k \neq j} \frac{m_j m_k (\vec{x}_k - \vec{x}_j)}{\|\vec{x}_k - \vec{x}_j\|^3}, \quad j = 1, \dots, n$$

where  $m_j$  are the masses, and  $\vec{x}_j = (x_j, y_j, z_j)$  are the positions of the objects.  $\gamma$  is the gravitational constant.

The (n = 2)-body problem was completely solved by Johann Bernoulli (1667–1748). But for  $n \ge 3$  no "easy" analytical solutions exist (in general). Even the (n = 3)-body problem can give rise to chaotic solutions.

Your task is to simulate the three-body system corresponding to the Sun–Earth–Moon system, using a numerical scheme (of your choice) of at least order 4. Run the simulation for a time-interval corresponding to at least 5 years (and make sure no cosmic disasters occur!)

Talk about any "engineering" decisions you made in your code in terms of normalizing parameters, step-lengths, and on-line or off-line error analysis.

Note that this problem is sort of vague and open-ended, and this description does not include all the information you need to solve the problem. Just like the real-world. :-)

## <sup>0</sup>Resources:

http://www.nasa.gov/

http://www.jpl.nasa.gov/

http://en.wikipedia.org/wiki/Sun

http://en.wikipedia.org/wiki/Earth

 $http://en.wikipedia.org/wiki/Earth's\_orbit$ 

http://en.wikipedia.org/wiki/Moon

 $http://en.wikipedia.org/wiki/Orbit_of\_the\_Moon$